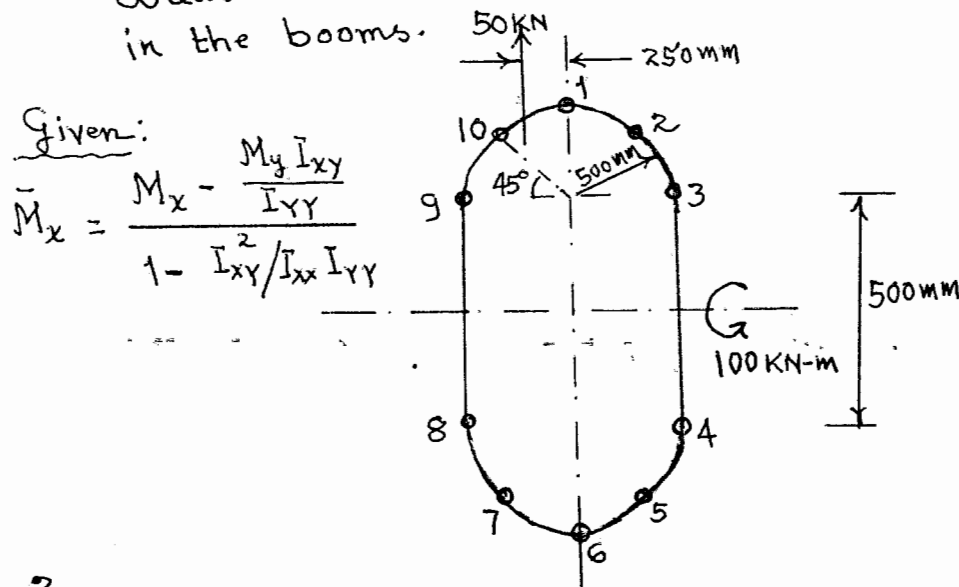


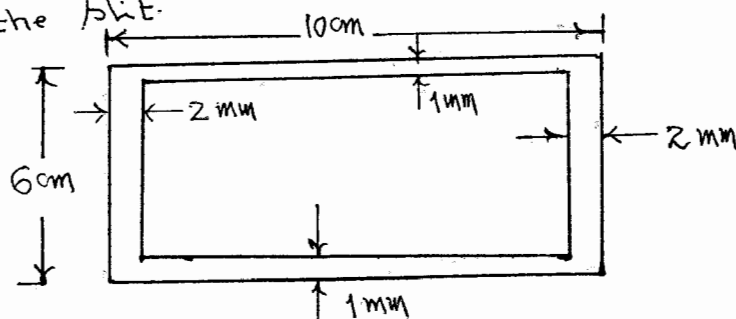
INDIAN INSTITUTE OF TECHNOLOGY

27.9.12 (A.N) 2 Hrs. P.M:100 No. of Students: 50
Autumn Sem, 2012 Aerospace Engg. AE-31009
3rd Yr. B.Tech (H) Aerospace Structural Analysis
Instruction: Answer to any Q. should be at one place only

- 1.
- (a) Explain the concept of symmetric and unsymmetric bending using neat sketches. (5)
- (b) The doubly symmetrical fuselage section shown has been idealized into an arrangement of direct stress carrying booms and shear stress carrying skin panels; the boom areas are all 150 mm^2 . Calculate the direct stresses in the booms. (20)



- 2.
- (a) Define Prandtl stress function in torsional problem, hence prove that the resultant forces acting on the ends of the bar vanish. Prove that $T = 2 \iint \phi(x,y) dx dy$ (8)
- (b) For the elasticity problem involving torsion, write down the mathematical relationships between St. Venant warping function, $\psi(x,y)$ and the Prandtl stress function, $\phi(x,y)$. (7)
- (c) A thin-walled rectangular tube shown, is subjected to torsion T . A slit is now made to the wall of the tube. Calculate the ratio of the torsional stiffness between the closed tube and the tube with the slit. (10)



3. A solid bar of elliptic cross-section is subjected to uniform torsion T . Using the Prandtl stress function solution technique, obtain the following:

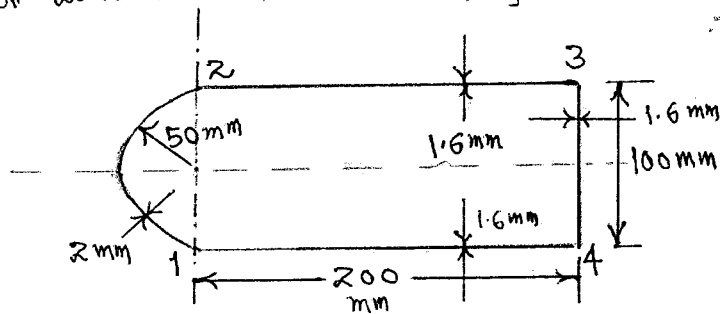
- the stress function $\phi(x, y)$ for the problem.
- the relationship between the applied torque and the rate of twist.
- hence, prove that $J_{\text{eff}}/I_p = (4a^2b^2)/(a^2+b^2)^2$
where, $I_p =$ Polar M. I. for an ellipse
- the value of the shear stresses at $x = \pm a, y = \pm b$
- Obtain the expression for the warping displacement, $w(x, y)$ and sketch the distribution.

5 X 5

4.

- For the single cell box section shown, assuming that the stresses are given by Bredth-Batho theory of torsion,
 - find the allowable torque on the tube so that the maximum shear stress on the cell wall does not exceed 140 MPa. Find the torsional rigidity of the tube. $G = 26 \text{ GPa}$ for all walls
 - find the torsional rigidity of the above section if G for wall 1-2 is 22 GPa, for all other walls $G = 26 \text{ GPa}$

(20)



- Explain: Number Tube with appropriate example.

(5)